

Recall: LAX Method:

$$\rho_i^{(n+1)} = \frac{1}{2} (\rho_{i+1}^{(n)} + \rho_{i-1}^{(n)}) - \frac{1}{2} C (\rho_{i+1}^{(n)} - \rho_{i-1}^{(n)})$$

$$C = \frac{\Delta t \cdot u}{\Delta x}$$

For linear advection equation, the flux is given by $f(\rho) = \rho u$

Rewrite the above using fluxes:

$$\rho_i^{(n+1)} = \frac{1}{2} (\rho_{i+1}^{(n)} + \rho_{i-1}^{(n)}) - \frac{\Delta t}{2 \Delta x} (f_{i+1}^{(n)} - f_{i-1}^{(n)})$$

(Method A)

Rewrite upwind scheme (1st order):

$$\rho_i^{(n+1)} = \rho_i^{(n)} - \frac{\Delta t}{\Delta x} (f_i^{(n)} - f_{i-1}^{(n)}) \quad \text{for } u \geq 0$$

$$\rho_i^{(n+1)} = \rho_i^{(n)} - \frac{\Delta t}{\Delta x} (f_{i+1}^{(n)} - f_i^{(n)}) \quad \text{for } u < 0$$

Now for 1-D hydrodynamics, 3 conservation equations:

1. Conservation of mass: $F = \rho u$

2. Conservation of momentum: $F = \rho u^2 + P$

3. Conservation of energy: $F = u(E + P)$

P is pressure, now let's use a "state vector"

$$\underline{U} = \begin{pmatrix} \rho \\ \rho u \\ E \end{pmatrix} \quad \underline{F}(\underline{U}) = \left[\rho u, \rho u^2 + P, u(E + P) \right]$$

velocity

$$\boxed{\frac{\partial \underline{U}}{\partial t} + \nabla \cdot \underline{F}(\underline{U}) = 0} \quad \text{now general}$$

$$\boxed{\frac{\partial \underline{U}}{\partial t} + \nabla \cdot \underline{F}(\underline{U}) = 0}$$

Very general
equation of
conservation of
"stuff"

3 equations, but 4 unknowns

$$\rho, u, P, E$$

$E =$ Kinetic Energy + Thermal Energy Density

$$= \frac{1}{2} \rho u^2 + \rho e$$

↳ specific internal energy

Still have 4 unknowns!

$$\rho, u, P, e$$

Ideal Gas: Equation of State

$$e = \frac{P}{(\gamma-1)\rho}$$

$$\gamma = \frac{f+2}{f}$$

where f is the
number of
degrees of
freedom

2-D Monoatomic Gas

$$\uparrow \rightarrow f=2 \quad \boxed{\gamma=2}$$

$$3-D: \gamma=5/3$$

in 1-D $\gamma=3$

$$\Rightarrow E = \frac{1}{2} \rho u^2 + \frac{1}{2} P \quad \text{for 1-D "gas"}$$

$$\underline{F}(\underline{U}) = \left[\rho u, \rho u^2 + P, u \left(\frac{1}{2} \rho u^2 + \frac{3}{2} P \right) \right]$$

for 1-D Monoatomic Gas

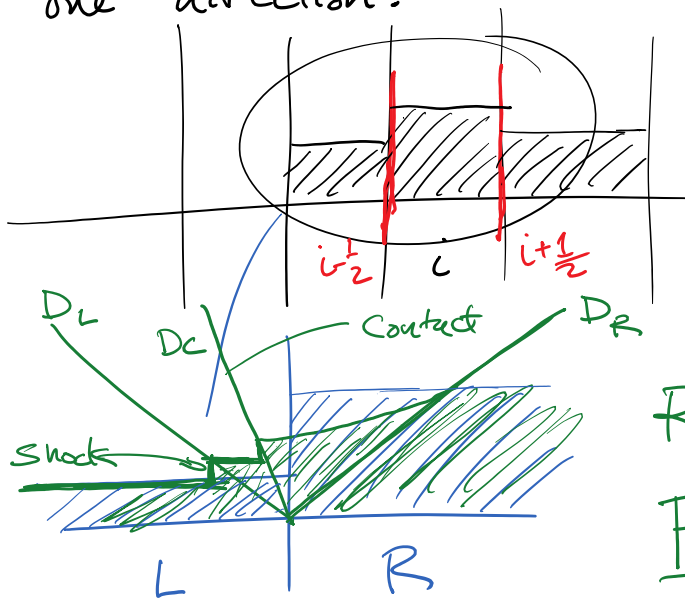
Method **[A]**: the original LAX Method

$$U_i^{(n+1)} = \frac{1}{2} (U_{i+1}^{(n)} + U_{i-1}^{(n)}) - \frac{\Delta t}{2\Delta x} (F_{i+1}^{(n)} - F_{i-1}^{(n)})$$

In finite volume let the values of U in each cell be the average of the "material" in the cell

$$\langle U \rangle_i^n = \frac{1}{\Delta x} \int_{x_{i-\frac{1}{2}}}^{x_{i+\frac{1}{2}}} u(x, t^n) dx$$

More complicated than for linear advection because information flows in more than one direction.



Gudonov

$$\frac{\langle u \rangle_i^{n+1} - \langle u \rangle_i^n}{\Delta t} + \frac{F_{i+\frac{1}{2}}^{n+1/2} - F_{i-\frac{1}{2}}^{n+1/2}}{\Delta x} = 0$$

Riemann Solver
 $\underline{F}(\underline{u}_L, \underline{u}_R)$

Approximate the Riemann Solver only 2 discontinuities, D_L, D_R

Also that $D_L = -D_R$!
 and the maximum value of the discontinuity.

$$D = |u| + c_s \quad \text{sound speed}$$

$$c_s = \sqrt{\gamma \frac{P}{\rho}}$$

$$D_{\max} = \max(D_L, D_R)$$

Lax-Friedrich Riemann Solver

$$F_{i-\frac{1}{2}}^n = \frac{1}{2} \left[F(\underline{u}_i^n) + F(\underline{u}_{i-1}^n) \right] - \frac{1}{2} D_{\max} \left[\underline{u}_i^n - \underline{u}_{i-1}^n \right]$$

Recall:

$$\underline{u}_i^{n+1} = \underline{u}_i^n - \frac{\Delta t}{\Delta x} \left[F_{i+\frac{1}{2}}^{(n)} - F_{i-\frac{1}{2}}^{(n)} \right]$$

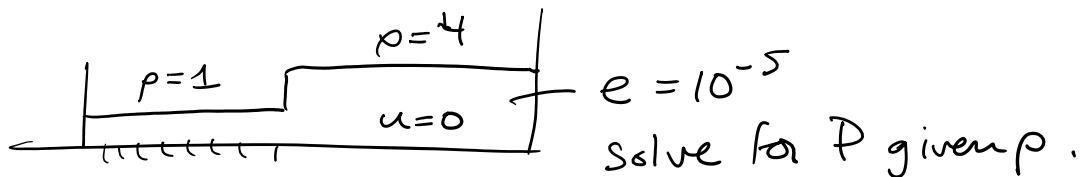
using n where we had $n+\frac{1}{2}$

\Rightarrow Method **B**

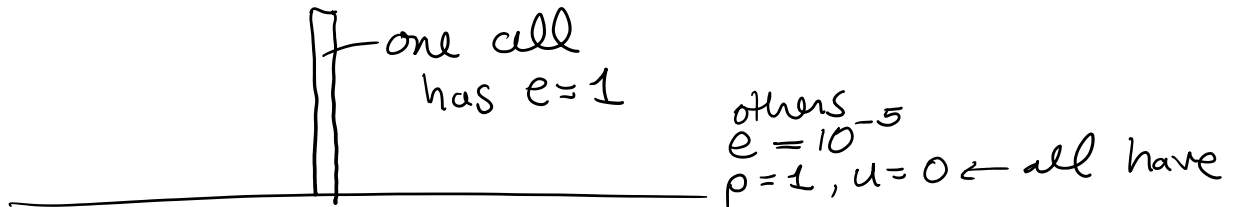
Method **A** does not involve a Riemann Solver, but Method **B** does.

"Shock Tube" Problems

Methods **A** and **B**



Sedov-Taylor Blast wave solution



$\Delta t = ?$ Can use D_{\max}

Courant Condition

$$D_{\max} < \frac{\Delta x}{\Delta t}$$